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the system includes electrical rotary type actuators the actuators operate mechanical linkages to open the gear and door uplocks and hydraulic valves vent v v and cut out v v

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all the flight control surfaces are hydraulically operated by actuators which receive electrical signals from the computers electrically controlled and hydraulically operated the rudder and the trimmable switch mode power converters

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horizontal stabilizer ths can also be mechanically controlled

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the a320 flight controls can be described as being electrically signaled controlled by full authority digital computers and hydraulically supplied some of them having hydromechanical backup

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on the other hand fly by wire aircraft usually integrate the yaw damper logic into the fly by wire commands sent to the actuators consider e g the airbus a320 and the boeing 777 airbus a320 electrical rudder control

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honeywell fti actuators are available for airbus a320 and boeing 737 747 757 and 777 aircraft flight controlsthe pitch trim actuator is a high performance direct power application actuator that responds to vehicle power inputs to apply load and rate to the vehicle s pitch surface based on system commands

electrical flight controls from airbus a320 330 340 to

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this is the so called electrical back up hydraulic actuator or ebha this actuator can be used to design flight control systems that survive the total loss of hydraulic power which is a significant advantage for a military transport aircraft particularly in the case of battle damage

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the a320 flight controls can be described as being electrically signaled controlled by full authority digital computers and hydraulically supplied some of them having hydromechanical backup

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yaw damper has four functions and controls the rudder via yaw damper actuators upon fmgc elac or fac orders the yaw damper provides dutch roll damping turn coordination in cruise engine failure compensation in auto flight yaw guidance order execution there are two yaw dampers in normal operation both are engaged but only one is active

flight controls what are the main components of the airbus

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in traditional aircraft the control surfaces like ailerons or elevators are linked to the control inputs used by the pilot e g the column this link can be direct cable and or amplified hydraulic or electric switch mode power converters design and analysis

actuator commanded by the pilot

avionics airbus control architecture where does the

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focusing particularly on the a320 family are the elacs and secs merely actuator controllers for the control surfaces or do they also carry the autopilot flight control algorithms or is the autopilot a separate entity which feeds desired control surface positions to the elacs and secs i e these latter are simply slaves to the autopilot

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while the aircraft was on a transit check for a scheduled flight airline s maintenance found an hydraulic leak from the engine n 1 inboard lower thrust reverser actuator in climb phase approximately 15 nm from the departing airport engine n 1 reverser got deployed

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the a380 a400m a350 features a mixed flight control actuation power source distribution associating electrically powered actuators with conventional flybywire hydraulic servocontrols

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the a320 aileron actuators have servo valves of course which when closed indeed allow no flow through them from pressure to return they additionally have mode valves which do provide a direct damped and narrow connection and it is this connection that allows for a slow and smooth aileron droop when unpowered

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the airbus a320 air conditioning system is fully automatic it continuously refreshes the air and maintains a selected temperature in three distinct zones the cockpit forward cabin and aft cabin each zone s temperature is controlled independently

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